

## **TITLE**

Electrostatic Switched Radiator For Space Based Thermal  
Control

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## **FIELD OF THE INVENTION**

The present invention relates to an improved device for  
controlling the effective emissivity of a surface by means  
of electrostatic attraction, which controls the thermal  
conductivity.

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## **BACKGROUND OF THE INVENTION**

Control of solar absorption and/or thermal  
emissivity is important for temperature control involving  
systems where radiation is the major heat control mechanism.  
Control of black body radiation and solar absorption, using  
a spectrally selective coating, will help control the  
temperature. But, when the heat load varies, active control  
of the thermal radiation is needed. Coolants have been used  
to conduct heat to an external radiator and can be  
controlled to block, or to be open, to piping. Louvers are  
another alternative that can be used to open or close. With  
a louver in one position, the exposed surface will have a  
high emissivity; alternately when the louvers is in the

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other position, the exposed surface will have a lower emissivity and will radiate less heat. When radiators are fixed, as in present art, options including heat pipes, heat pump systems, capillary pump looped heat pipes and louvers  
5 can be effective but are expensive, heavy and bulky.

Electrostatic forces have been used previously in various applications.

U.S. Pat. No. 4665463 (1986) to Ward *et al.* describes an electrostatic chuck for holding a semiconductor wafer,  
10 comprising a dielectric layer on a supporting electrode. A potential is applied between the wafer and the electrode and the dielectric is loaded with thermally conductive material to improve dissipation of heat generated in the wafer during a processing treatment such as exposure to an electron beam.

15 U.S. Pat. No. 4771730 (1987) to Tezuka *et al.* describes a vacuum processing apparatus with a vacuum vessel within which a work to be processed is drawn and held fixed on a specimen table by an electrode functioning doubly as an electrostatic chuck, to which is connected a gas feeding  
20 pipe for feeding a gas affording good heat transmission between the mutually contacting surfaces of the work and the electrode to control the temperature of the work.

What is needed is a smaller, less expensive, flexible, lighter weight, higher performance, and more reliable solution. The present invention solves these problems with use of an electrostatic switched radiator.

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### SUMMARY OF THE INVENTION

The main aspect of the present invention is to provide an electrically switched radiator for space based thermal control by means of an electrostatic hold-down or release of a thin composite film to control inner compartment craft/spacecraft temperature.

Another aspect of the present invention is to provide thermal control by producing a large change in effective emissivity when switching the device from the "off" (non-radiating) to the "on" (radiating) stage.

Another aspect of the present invention is to provide a high emissivity composite film to control craft/spacecraft skin temperature.

Another aspect of the present invention is to provide a means for switching the effective emissivity from a low to a high value and visa versa via contact/non-contact with a surface to be cooled.

Another aspect of the present invention is to provide for a thin composite film which is flexible for good contact with the outer skin of the craft.

Another aspect of the present invention is to provide a  
5 low cost, low weight, high performance, high reliability and small size electrostatically controlled radiator for thermal control of craft/spacecraft temperatures.

Other aspects of this invention will appear from the following description and appended claims, reference being  
10 made to the accompanying drawings forming a part of this specification wherein like reference characters designate corresponding parts in the several views.

The present invention utilizes a high emissivity composite film to control craft/spacecraft skin temperature.  
15 The electrostatic hold-down switches the mode of heat transfer from the craft/spacecraft skin from a conduction mode to a radiation mode and back. The device is referred to as the "Electrostatic Switched Radiator" or ESR. The ESR is very lightweight and has demonstrated with experimentation  
20 the capability to switch the emissivity from below 0.1 to above 0.95. Emissivity is simply the ratio of the actual emitted radiance to that of an ideal blackbody. Emissivity ranges from 0 to 1 where 1 would be a blackbody. Emissivity

can also vary with wavelength for any particular substance. For example, the emissivity for a water droplet decreases as the wavelength decreases.

5 The ESR construction is simple and lightweight. It consists of a thin polymer film, with a weight of less than a few hundred grams/m<sup>2</sup>. The film can be anchored to the craft/spacecraft at the edges. The cover film consists of a high dielectric constant insulator with a good dielectric strength and is coated on its outer surface with an electrically conductive thin layer. The outer surface of the ESR is constructed to have a very high emissivity, ideally with low visible absorbance. This combination can be achieved with an appropriate paint or, for better performance, a multi-layer thin film designed for very low visible absorbance and high emissivity. The top surface or "skin" of the craft/spacecraft, to which the ESR will be "in contact" or "not in contact" should have a very low emissivity, i.e. sputtered gold.

20 Basic heat control is simple and highly effective. When the ESR is turned on, the emitting surface is in good thermal contact with an inner surface skin (such as the outer skin of a spacecraft). This results in good heat conduction between the craft surface (skin) and the ESR such

that the emitting surface of the film is at the craft skin temperature. The emitting ESR surface radiates at the "skin" temperature (high emissivity state). When the ESR is turned off, the film moves away from the skin (is not in contact)

5 and the heat flow is only radiation from the inner surface  
skin (low emissivity). Thus, once it reaches equilibrium,  
the film can only radiate the heat it absorbs which is  
limited to radiation from the inner surface skin. The inner  
surface skin is fabricated with a low emissivity and thus in  
10 the released state, the outer skin emissivity doesn't  
change, however it's temperature drops and the result is a  
drop in the radiated energy. This approach avoids the need  
for an infrared (IR) transparent conductor, which is always  
difficult since transparent conductors (wide band gap  
15 semiconductors with high electron concentration) have  
significant absorption in the IR.

The ESR requires minimal material requirements and the system is compatible with conventional paints and coatings for full utilization in low solar absorbance, high emissivity coatings. Typical films could consist of an outside coating of copper with only 1000 angstrom to 25-50 micron thickness. A sputtered metal on a polymer will improve hold-down via a more pliable structure and allow

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operation at a lower applied voltage. The film itself is an insulator such as polyimide (Kapton) and requires a high dielectric constant, high thermal conductivity, and high dielectric strength properties. Other films such as Kynar are also alternatives. The ESR operates as a high quality capacitor with a dielectric (film) between two layers of metal (film metallic coating and craft metallic coating). The surface area (radiating area) of the ESR is calculated to dissipate heat needed to control the internal temperature of the vehicle or craft. Surface area is selected as a function of heat generated in order to determine the amount of heat to be radiated. The ESR can be subdivided into sectional areas depending on design requirements. The skin or area of contact, such as the outer skin of a craft, is required to be metallic or metallic coated (typically aluminum). Typical internal craft temperatures are often controlled around 300 degrees Kelvin (Room Temperature). The heat generated inside will raise the internal temperature and the ESR will dissipate the heat to control the internal temperature to room temperature.

Switching a DC voltage controls the "on" versus the "off" state of the ESR. The ESR will operate effectively with moderate levels of DC voltage (typically 100-500 VDC).





emissivity levels of lower than 0.1. Thus devices with these characteristics would generate a change of greater than 0.8 in switched emissivity levels.

Actual test measurements of a working device contained  
5 within a vacuum bell jar were performed using a copper block with an area consisting of a flat black painted strip as the high emissivity reference, an area of bare copper as a low emissivity reference, and an area with the ESR. Measurements were taken using an Inframetrics 625 imager, which is a  
10 camera sensitive from 8 to 14 microns. Test results showed that, within the limits of the test setup, the "on" state of the ESR was approximately the same as the black painted substrate and the "off" state of the ESR was approximately the same energy as the bare copper. The black painted area  
15 was estimated to have an emissivity of approximately 0.95 and the bare copper an emissivity of less than 0.01. Thus, within the limits of the experimental measurements, the high emissivity "on" state (with electrostatic hold-down) was shown to have the same emissivity as the black painted strip  
20 and the low emissivity "off" state (with no electrostatic hold-down) was shown to have the same emissivity as the bare copper. This test showed that electrostatic hold-down would insure good thermal contact and that a vacuum system could

produce a sufficiently low pressure to eliminate or minimize thermal conduction from the air.

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Since this measurement is sensitive to the wavelength of the detector, additional measurements were made in which measurements of the heat loss were used to determine the emissivity. With this measurement, the sample is placed on a thermal control plate and the heat loss is measured by measuring the power required to maintain temperature and is basically a calorimetric approach. The heat loss with ESR switching gives a very accurate measure of a change of emissivity. Absolute values of emissivity require a calibrated sample, which used a sputtered gold film as the "zero" emissivity point and a black paint for the high emissivity value ( $\epsilon \sim 0.9$ ).

Fig. 3 (below) shows measured results for a sample consisting of a cover film with a thin aluminum film. For this measurement, the voltage was applied while the sample was warm and showed a change of the effective emissivity of 0.74. This test showed that electrostatic hold-down would insure good thermal contact.

### BRIEF DESCRIPTION OF THE DRAWINGS

Fig. 1A, 1B is a depiction of the ESR in the "OFF" and  
5 the "ON" states, respectively.

Fig. 2A is a cross-sectional view of the ESR with a  
voltage source and switch and the ESR in the "OFF" position.

Fig. 2B is a cross-sectional view of the ESR with a  
voltage source and switch and the ESR in the "ON" position.

10 Fig. 3 is a graph of power input measurements of a thin  
metallized ESR.

Fig. 4 is a depiction of a craft with two ESRs  
attached, one in the "ON" and one in the "OFF" position.

Before explaining the disclosed embodiment of the  
15 present invention in detail, it is to be understood that the  
invention is not limited in its application to the details  
of the particular arrangement shown, since the invention is  
capable of other embodiments. Also, the terminology used  
herein is for the purpose of description and not of  
20 limitation.

### Detailed Description of Drawings

Fig. 1A is a depiction of the ESR 1000 in the "OFF" state. The outer skin 100 of a craft radiates heat via radiation **Rs**. Thermal gap **G** acts as an insulator. Energy is radiated **Rs** by the outer skin 100, which has a low surface emissivity. Heat is absorbed by the composite film 102 consisting of the dielectric 101 and the thin metallic surface coat 104. In this "OFF" position, the heat loss from the outer skin 100 is the energy lost from radiation emitted **Rs** minus the energy absorbed from the reflected radiated energy **Ra** of the composite film 102 (dielectric 101 and thin metallic surface coat 104) and minus the energy absorbed from the radiated energy **Rb** by the surface of the composite film 102 (dielectric 101 and metallic surface coat 104) at an unknown temperature. The composite film 102 (dielectric 101 and outer thin metallic surface coat 104) thus emits reflected radiated energy **Ra** back to the skin 100 and radiated energy **Rb** from the surface temperature. Energy is also transferred into space by radiated energy **Rc** from the high emissivity surface of metallic surface coating 104. A steady state temperature will occur when the temperature of the dielectric 101 and metallic surface coat 104 reaches a

temperature such that it's energy absorbed is equal to its energy radiated.

Fig. 1B is a depiction of the ESR **1000** in the "ON" state. In this state, the outer skin **100** of a craft is in close contact **103** (thermal contact) with the thin film dielectric **101** and its metallic coating **102**. In this "ON" state heat is conducted away from the outer skin **100** of the craft at a very high level of efficiency and transferred into space by radiation **Rc** off the metallic surface **104** which is designed at a high emissivity.

Fig. 2A is a cross-sectional view of the ESR **1000** with a voltage source **204** and switch **205** open and thus the ESR **1000** in the "OFF" position. The metallic surface coat **104** and thin film dielectric **101** are separated from the outer skin **100** of the craft by a thermal gap **G**. Non-conductive hinges **201** could be used to facilitate the separation, as could a piezoelectric strip. A DC circuit consists of a DC voltage source **204**, a switch **205** in the open position, an outer skin contact point **207**, a connector **202** and insulator **206** and a wire contact **203** to the metallic surface coat **104**.

Fig. 2B is a cross-sectional view of the ESR **1000** with a voltage source **204** and switch **205** in the "ON" position. In this position the thermal gap **G** is basically zero. Thus, the metallic surface coat **104** and thin film dielectric **101** are in direct contact with the outer skin **100** of the craft. Various methods are possible to insure that this gap **G** is sufficient to limit the heat transfer. This can include non-conductive hinges **201** as shown, which are designed to collapse. Normal elasticity of the cover film when stretched and mounted at the ends may be sufficient to insure such a gap. A piezoelectric strip attached to the cover film as previously described could also be used. In this "ON" position the DC voltage source **204** pulls the thin film dielectric **101** and its metallic surface coat **104** into thermal contact with the outer skin **100**. The ESR **1000** transfers heat from the outer skin **100** and radiates the heat into space by the high emissivity surface of the metallic surface coat **104**. As also shown in Fig. 2A, is a DC circuit consisting of a DC voltage source **204**, a switch **205** in the closed position, an outer skin contact point **207**, a connector **202** and insulator **206** and a wire contact **203** to the metallic surface coat **104**.

Fig. 3 is a graph of power input measurements of a thin metallized ESR as was discussed above; the sample tested having a change of emissivity of 0.74. When the ESR was switched from a low "off" state **300** to a high "on" state

5 **301** it can be seen that power requirements decreased

Fig. 4 is a depiction of a craft **400** with two ESRs attached, one in the "ON" position **405** and one in the "OFF" position **406**. Energy released within a craft can come from electronics **403** and/or human occupants **404** which would  
10 increase the internal craft temperature if no control were present. Electronic switching sensors **401**, **402** would allow for electrostatic switching of the ESRs **405**, **406** by allowing a voltage to be "on" or "off". Contact points for the electronic sensor **401** are at the outer skin **408** and at the  
15 metallic surface **407** of the ESR **406**, which is shown in the "OFF" or non-contact mode. Thus, little or no energy is radiated into space **S** as previously discussed. Contact points for the electronic sensor **402** are at the outer skin **410** and at the metallic surface **409** of the ESR **405**, which is  
20 shown in the "ON" or contact mode. Thus, energy **Rc** is radiated into space **S** as previously discussed.

Although the present invention has been described with reference to preferred embodiments, numerous modifications and variations can be made and still the result will come within the scope of the invention. No limitation with  
5 respect to the specific embodiments disclosed herein is intended or should be inferred.